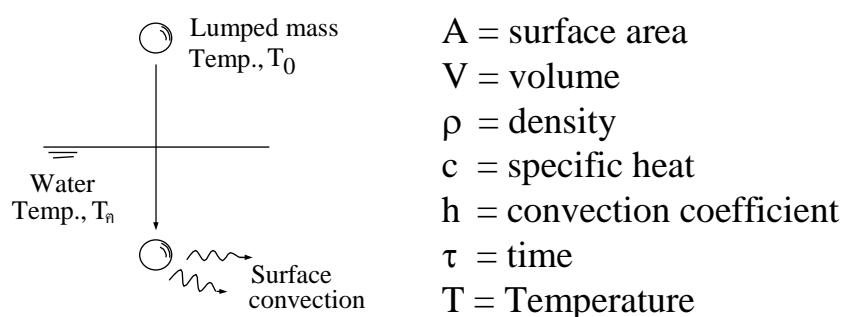


METHOD OF WEIGHTED RESIDUALS

METHOD OF WEIGHTED RESIDUALS

To understand the Method of Weighted Residuals (MWR), let's consider the example of transient thermal response of a lumped mass. Given,



Want: Lumped mass temperature, $T = T(\tau) = ?$

TRANSIENT RESPONSE OF LUMPED MASS

Conservation of energy:

$$\text{Energy in} - \text{Energy out} = \text{Energy stored}$$

$$0 - h A (T - T_{\infty}) = \rho c V \frac{\partial T}{\partial \tau}$$

$$\rho c V \frac{\partial T}{\partial \tau} + h A (T - T_{\infty}) = 0$$

with initial condition: $T(\tau = 0) = T_0$

To easily solve this ODE, let's denote,

$$\text{Non-dimensional temperature } \bar{x} = \frac{T - T_{\infty}}{T_0 - T_{\infty}}$$

$$\text{Non-dimensional time } \bar{t} = \frac{h A \tau}{\rho c V}$$

then the ODE & IC become,

TRANSIENT RESPONSE OF LUMPED MASS

ODE: $\frac{d\bar{x}}{dt} + \bar{x} = 0$

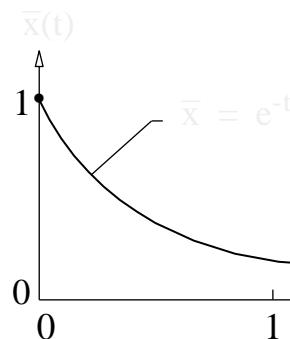
IC: $\bar{x}(t = 0) = 1$

To solve for exact solution, $\bar{x}(t)$, we separate the variables, $\frac{d\bar{x}}{\bar{x}} = -dt$

then integrate $\ln \bar{x} = -t + B$

where B is the integrating constant which is zero after applying the initial condition. Thus, the exact solution is, $\bar{x} = \bar{x}(t) = e^{-t}$

TRANSIENT RESPONSE OF LUMPED MASS



Exact solution:

$$\bar{x} = \bar{x}(t) = e^{-t}$$

approaches zero
as $t \rightarrow \infty$

Now, if we don't know the exact solution, and we want to use the method of weighted residuals to find approximate solution for $0 < t < 1$.

METHOD OF WEIGHTED RESIDUALS

Assume an approximate solution in the form,

$$x(t) = 1 + C_1 t + C_2 t^2$$

where C_1 and C_2 are constants to be determined. Note that if we substitute this approx. sol. into the diff. eq.

$$\frac{dx}{dt} + x \neq 0 \quad \text{in fact} \quad = R$$

$$\text{i.e., } R = (0 + C_1 + 2C_2 t) + (1 + C_1 t + C_2 t^2)$$

$$\text{or, } R = R(t) = 1 + C_1(1 + t) + C_2(2t + t^2)$$

So, the idea is to determine C_1 and C_2 such that the Residual $R \rightarrow 0$.

METHOD OF WEIGHTED RESIDUALS

Four approaches:

- (1) Point collocation Set residuals at selected points to be zero.
- (2) Subdomain collocation Set areas under the residual curve to be zero.
- (3) Galerkin Include weighting functions. Widely used in FE method.
- (4) Least squares Square the residual with minimization process.

POINT COLLOCATION

Set the residuals at selected points to be zero, i.e.,

$$R(t_i) = 0 \quad i = 1, 2, \dots, \text{no. of unknowns}$$

Since there are two unknowns of C_1 and C_2 , thus we need 2

e.q.s. of, $R(t_1) = 0$ and $R(t_2) = 0$

where t_1 and t_2 should be in the range of consider 0 to 1.

E.g., select,

$$t_1 = \frac{1}{3}; \quad R\left(\frac{1}{3}\right) = 1 + C_1\left(1 + \frac{1}{3}\right) + C_2\left(2\left(\frac{1}{3}\right) + \left(\frac{1}{3}\right)^2\right) = 0$$

$$t_2 = \frac{2}{3}; \quad R\left(\frac{2}{3}\right) = 1 + C_1\left(1 + \frac{2}{3}\right) + C_2\left(2\left(\frac{2}{3}\right) + \left(\frac{2}{3}\right)^2\right) = 0$$

Solve to get, $C_1 = -0.9310$ and $C_2 = 0.3103$

Then, $x(t) = 1 - 0.9310 t + 0.3103 t^2$

SUBDOMAIN COLLOCATION

Set areas under the residual curve to be zero.

Since there are 2 unknowns, thus we need 2 eqs., e.g.

$$\int_0^{t_1} R(t) dt = 0 \quad \text{and} \quad \int_{t_1}^1 R(t) dt = 0$$

If we select $t_1 = \frac{1}{2}$, then,

$$\int_0^{1/2} R(t) dt; \quad \frac{1}{2} + \frac{5}{8}C_1 + \frac{7}{24}C_2 = 0$$

$$\int_{1/2}^1 R(t) dt; \quad \frac{1}{2} + \frac{7}{8}C_1 + \frac{25}{24}C_2 = 0$$

Solve to get, $C_1 = -0.9474$ and $C_2 = 0.3158$

Then, the approximate solution is,

$$x(t) = 1 - 0.9474 t + 0.3158 t^2$$

GALERKIN APPROACH

Multiply the residual by some weighting functions, perform integration over the entire domain, and set to be zero.

$$\int_0^1 R(t) W_i(t) dt = 0 \quad i = 1, 2, \dots, \text{No. of unknowns}$$

where $W_i(t)$ is the weighting functions which are normally selected as the terms that weight with the unknowns, i.e.,

$$x(t) = 1 + C_1 \underbrace{\frac{t}{W_1}} + C_2 \underbrace{\frac{t^2}{W_2}}$$

here, we use $W_1 = t$ and $W_2 = t^2$, then substitute into the above eq. to get,

GALERKIN APPROACH

$$\int_0^1 R(t) t \, dt = 0; \quad \frac{1}{2} + \frac{5}{6}C_1 + \frac{11}{12}C_2 = 0$$

$$\int_0^1 R(t) t^2 \, dt = 0; \quad \frac{1}{3} + \frac{7}{12}C_1 + \frac{7}{10}C_2 = 0$$

Solve to get, $C_1 = -0.9143$ and $C_2 = 0.2857$

Thus, the approximate solution is,

$$x(t) = 1 - 0.9143 t + 0.2857 t^2$$

Note: We will use this approach to derive FE eqs. later. Since distribution of element variable is,

$$\phi = \underbrace{\frac{N_1}{W_1}}_{\phi_1} \phi_1 + \underbrace{\frac{N_2}{W_2}}_{\phi_2} \phi_2$$

Here, the nodal unknowns ϕ_1 and ϕ_2 are similar to C_1 and C_2 . The element interpolation functions N_1 and N_2 will be selected as the weighting functions.

LEAST SQUARES APPROACH

Square the residual, perform integration over entire domain, and minimize with respect to the unknowns, i.e.,

$$\frac{\partial}{\partial C_i} \int_0^1 R^2(t) \, dt = 0 \quad i = 1, 2, \dots, \text{No. of unknowns}$$

Since there are 2 unknowns, thus we have 2 eqs.,

$$\frac{\partial}{\partial C_1} \int_0^1 R^2(t) \, dt = 0; \quad \frac{3}{2} + \frac{7}{3}C_1 + \frac{9}{4}C_2 = 0$$

$$\frac{\partial}{\partial C_2} \int_0^1 R^2(t) \, dt = 0; \quad \frac{4}{3} + \frac{9}{4}C_1 + \frac{38}{15}C_2 = 0$$

Solve to get, $C_1 = -0.9427$ and $C_2 = 0.3110$

Then, the approximate solution is,

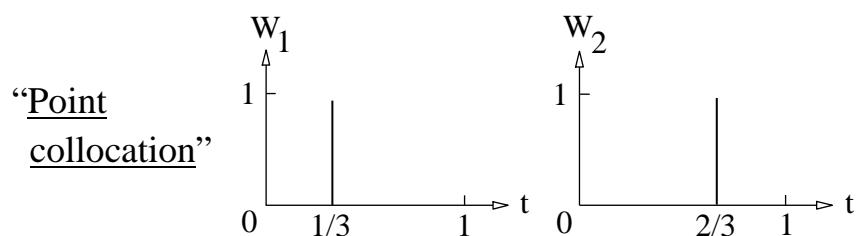
$$x(t) = 1 - 0.9427 t + 0.3110 t^2$$

METHOD OF WEIGHTED RESIDUALS

In conclusion, these four approaches can be written in the form,

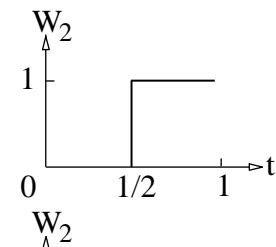
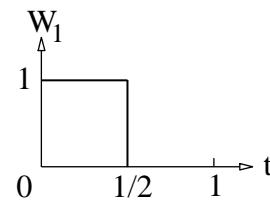
$$\int_0^1 R(t) W_i(t) dt = 0$$

where W_i are the weighting functions that can be described as follows,

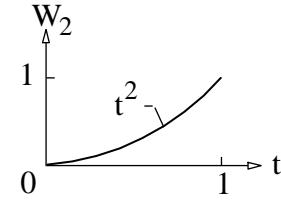
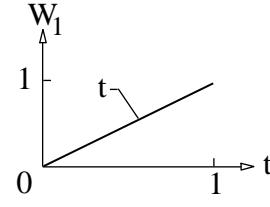


METHOD OF WEIGHTED RESIDUALS

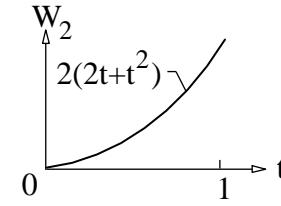
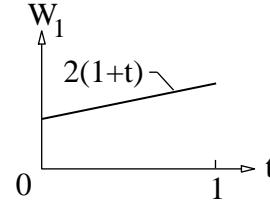
"Subdomain collocation"



"Galerkin"

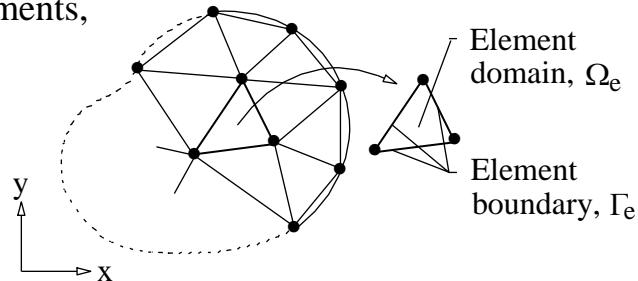


"Least squares"



GENERAL PROCEDURE OF MWR

Step 1: Discretize the given domain into a number of elements,



Then identify the governing differential eq(s) of the problem,

$$L(\bar{\phi}) = 0$$

where L denotes the differential operator and $\bar{\phi}$ is the exact solution.

GENERAL PROCEDURE OF MWR

Step 2: Assume element approximate solution in the form,

$$\phi = \phi(x, y) = \sum_{i=1}^m N_i \phi_i = [N] \{ \phi \}$$

where $\phi = \phi(x, y)$ = element approximate solution

$N_i = N_i(x, y)$ = element interpolation functions

ϕ_i = element nodal unknowns

m = number of element nodes

Step 3: Apply the method of weighted residuals for a typical element. If we substitute the approximate solution, $\phi(x, y)$, into the governing diff. eq.,

$$L(\phi) \neq 0 \text{ in general, but } L(\phi) = \text{Error} = R$$

GENERAL PROCEDURE OF MWR

Here R is called the Residual, i.e.,

$$R = L(\phi) = L(\{N_i\} \{\phi\}) = L\left(\sum_{i=1}^m N_i \phi_i\right)$$

Then apply the method of weighted residuals (Galerkin approach):

Multiply the Residual R by some weighting functions, perform integration over element domain, and set to be zero.

$$\int_{\Omega^{(e)}} W_i R d\Omega = 0 \quad i = 1, 2, \dots, m$$

Note : If $W_i = N_i$ \Rightarrow called “Bubnov-Galerkin” approach
 $W_i \neq N_i$ \Rightarrow called “Petrov-Galerkin” approach

GENERAL PROCEDURE OF MWR

Step 4: Perform integration by parts (or use Gauss's theorem).

$$\begin{aligned} \int_{\Omega^{(e)}} W_i R d\Omega &= \int_{\Omega^{(e)}} W_i L\left(\sum_{i=1}^m N_i \phi_i\right) d\Omega \\ &= \underbrace{\int_{\Omega^{(e)}} (W_i, N_i, \phi_i) d\Omega}_{\substack{\text{Associated with} \\ \text{element domain}, \Omega^{(e)}}} + \underbrace{\int_{\Gamma^{(e)}} (W_i, N_i, \phi_i) d\Gamma}_{\substack{\text{Associated with} \\ \text{element boundary}, \Gamma^{(e)}}} = 0 \end{aligned}$$

Step 5: Apply boundary conditions on the element boundary integral term $\int_{\Gamma^{(e)}}$ as needed.

GENERAL PROCEDURE OF MWR

Step 6: Write the element eqs. in matrix form,

$$[K] \begin{Bmatrix} \{\phi\} \\ (mxm) \end{Bmatrix} = \begin{Bmatrix} \{F\} \\ (mx1) \end{Bmatrix}$$

where $[K]$ = Element stiffness matrix

$\{\phi\}$ = Vector of element nodal unknowns

$\{F\}$ = Vector of element nodal loads

Then assemble all element eqs., apply all BC's on the system eqs., and solve for nodal unknowns.

METHOD OF WEIGHTED RESIDUALS

Objective: Derive finite element equations from governing differential equation.

Simplest example is 1-D Poisson's eq.:

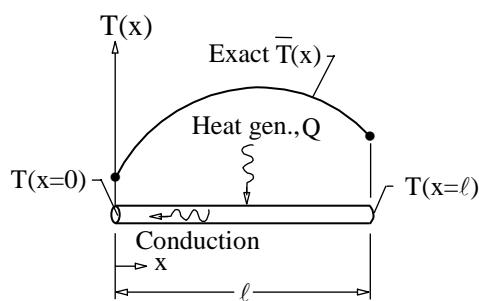
$$\frac{d^2 u}{dx^2} = f(x)$$

where $u = u(x)$. Such differential eq. may represent the problems of,

- Bar with its own weight
- 1-D viscous flow
- 1-D heat conduction with internal heat generation

etc.

CONDUCTION WITH HEAT GENERATION



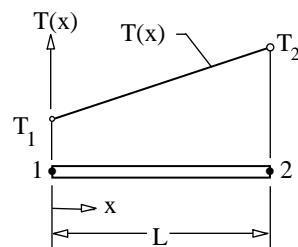
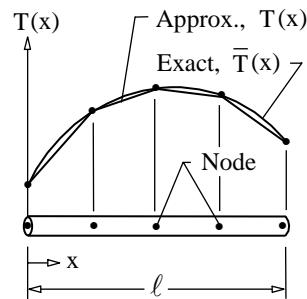
Differential equation:

$$kA \frac{d^2\bar{T}}{dx^2} = -QA$$

where k is thermal conductivity and A is cross-sectional area.

Boundary conditions: $T(x = 0) = T_0$ and $T(x = l) = T_l$

DISCRETIZATION & INTERPOLATION FUNC.



Assume linear element temperature distribution,

$$T(x) = ax + b$$

$$\text{At } x = 0; \quad T(x = 0) = T_1 = b$$

$$\text{At } x = L; \quad T(x = L) = T_2 = aL + b$$

$$\text{Then } a = \frac{T_2 - T_1}{L} \quad \text{and} \quad b = T_2$$

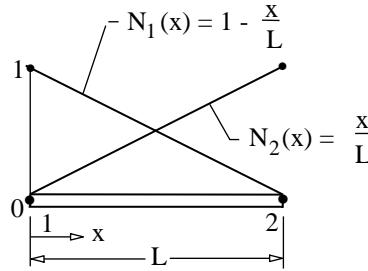
ELEMENT INTERPOATION FUNCTIONS

$$\text{Then } T(x) = \left(1 - \frac{x}{L}\right)T_1 + \left(\frac{x}{L}\right)T_2$$

which can be written in the form,

$$T(x) = N_1 T_1 + N_2 T_2 = \begin{bmatrix} N_1 & N_2 \end{bmatrix} \begin{bmatrix} T_1 \\ T_2 \end{bmatrix} = \begin{bmatrix} N \\ T \end{bmatrix}_{(1 \times 2)} \begin{bmatrix} T \end{bmatrix}_{(2 \times 1)}$$

where the element interpolation functions,



$$\begin{aligned} N_1 &= 1 - \frac{x}{L} \\ N_2 &= \frac{x}{L} \end{aligned}$$

with properties of,

$$N_i = \begin{cases} 1 & \text{at node } i \\ 0 & \text{at other node} \end{cases}$$

METHOD OF WEIGHTED RESIDUALS

If we can solve for exact solution, \bar{T} , and substitute into the differential eq., then RHS will be identically zero, i.e.,

$$kA \frac{d^2\bar{T}}{dx^2} + QA = 0$$

However, in general, we don't know exact solution. Thus, if we substitute the approximate solution, T , RHS will not be zero, but will be equal to a Residual, R :

$$kA \frac{d^2T}{dx^2} + QA = R$$

The idea is thus to minimize R so that error is minimum.

METHOD OF WEIGHTED RESIDUALS

The method of weighted residuals is to multiply R by some weighting functions W_i , perform integration over element domain, and set to zero, i.e.:

$$\int_0^L W_i R \, dx = 0$$

Since the element has two unknowns, thus we need 2 eqs., i.e., $i = 1, 2$. Substitute the residual to get,

$$\int_0^L W_i \left(kA \frac{d^2 T}{dx^2} + QA \right) dx = 0$$

Expand,

$$\int_0^L W_i k A \frac{d^2 T}{dx^2} dx + \int_0^L W_i Q A dx = 0 \quad i = 1, 2$$

METHOD OF WEIGHTED RESIDUALS

Apply integration by parts to the first integral term in order to produce the boundary term, i.e.,

$$\int_0^L \underbrace{W_i}_{u} \underbrace{k A \frac{d^2 T}{dx^2}}_{dv} dx = W_i k A \frac{dT}{dx} \Big|_0^L - \int_0^L k A \frac{dT}{dx} \frac{dW_i}{dx} dx$$

by using the formula, $\int_0^L u dv = u v \Big|_0^L - \int_0^L v du$

where $u = W_i$ then $du = \frac{dW_i}{dx} dx$

and $dv = k A \frac{d^2 T}{dx^2} dx$ then $v = k A \frac{dT}{dx}$

Then, the element eqs. becomes,

METHOD OF WEIGHTED RESIDUALS

$$\int_0^L k A \frac{dW_i}{dx} \frac{dT}{dx} dx = W_i k A \left. \frac{dT}{dx} \right|_0^L + \int_0^L W_i Q A dx$$

Recall $T = [N_1 \ N_2] \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix}$ then $\frac{dT}{dx} = \left[\frac{dN_1}{dx} \ \frac{dN_2}{dx} \right] \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix}$

Substitute and write out for $i = 1, 2$ to get 2 equations as,

$$i=1; \int_0^L k A \frac{dW_1}{dx} \left[\frac{dN_1}{dx} \ \frac{dN_2}{dx} \right] \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} dx = W_1 k A \left. \frac{dT}{dx} \right|_0^L + \int_0^L W_1 Q A dx$$

$$i=2; \int_0^L k A \frac{dW_2}{dx} \left[\frac{dN_1}{dx} \ \frac{dN_2}{dx} \right] \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} dx = W_2 k A \left. \frac{dT}{dx} \right|_0^L + \int_0^L W_2 Q A dx$$

METHOD OF WEIGHTED RESIDUALS

For standard FE., we select $W_i = N_i$ which is known as Bubnov-Galerkin approach, we will get element equations in the form,

$$\begin{aligned} & \int_0^L k A \left\{ \begin{array}{c} \frac{dN_1}{dx} \\ \frac{dN_2}{dx} \end{array} \right\} \left[\frac{dN_1}{dx} \ \frac{dN_2}{dx} \right] \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} dx \\ &= \left(\begin{Bmatrix} N_1 \\ N_2 \end{Bmatrix} k A \frac{dT}{dx} \right) \Big|_0^L + \int_0^L \begin{Bmatrix} N_1 \\ N_2 \end{Bmatrix} Q A dx \end{aligned}$$

METHOD OF WEIGHTED RESIDUALS

Or, in short: $[K_C]\{T\} = \{Q_C\} + \{Q_Q\}$

where $[K_C]$ = Element conduction matrix

$\{T\}$ = Vector of element nodal temperatures

$\{Q_C\}$ = Vector of element nodal heat fluxes

$\{Q_Q\}$ = Vector of element nodal heat generations

METHOD OF WEIGHTED RESIDUALS

$$\text{Conduction matrix } [K_C] = \int_0^L k A \left\{ \frac{dN_1}{dx} \right\} \left[\frac{dN_1}{dx} \quad \frac{dN_2}{dx} \right] dx$$

$$\text{Since } N_1 = 1 - \frac{x}{L} \quad \text{and} \quad N_2 = \frac{x}{L}$$

$$\text{then } \frac{dN_1}{dx} = -\frac{1}{L} \quad \text{and} \quad \frac{dN_2}{dx} = \frac{1}{L}$$

$$\text{Substitute } [K_C] = \int_0^L k A \begin{Bmatrix} -\frac{1}{L} \\ \frac{1}{L} \\ \frac{1}{L} \end{Bmatrix} \begin{Bmatrix} -\frac{1}{L} & \frac{1}{L} \end{Bmatrix} dx = \int_0^L \frac{k A}{L^2} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} dx$$

If thermal conductivity k & cross-sectional area A are constant,
then

$$[K_C] = \frac{k A}{L^2} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

ELEMENT LOAD VECTORS

Load vector due to internal heat generation

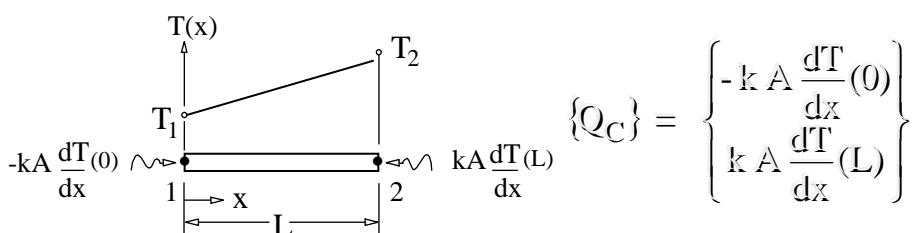
$$\{Q_Q\} = \int_0^L \begin{Bmatrix} N_1 \\ N_2 \end{Bmatrix} Q A dx = \int_0^L \begin{Bmatrix} 1 - \frac{x}{L} \\ \frac{x}{L} \end{Bmatrix} Q A dx = Q A L \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix}$$

Load vector due to nodal heat fluxes

$$\begin{aligned} \{Q_C\} &= \left(\begin{Bmatrix} N_1 \\ N_2 \end{Bmatrix} k A \frac{dT}{dx} \right) \Big|_0^L = \left\{ \begin{array}{l} \left(N_1 k A \frac{dT}{dx} \right) \Big|_0^L \\ \left(N_2 k A \frac{dT}{dx} \right) \Big|_0^L \end{array} \right\} \\ &= \left\{ \begin{array}{l} N_1(L) k A \frac{dT}{dx}(L) - N_1(0) k A \frac{dT}{dx}(0) \\ N_2(L) k A \frac{dT}{dx}(L) - N_2(0) k A \frac{dT}{dx}(0) \end{array} \right\} \end{aligned}$$

But $N_1(0) = 1, N_1(L) = 0, N_2(0) = 0, N_2(L) = 1$, thus

ELEMENT LOAD VECTORS



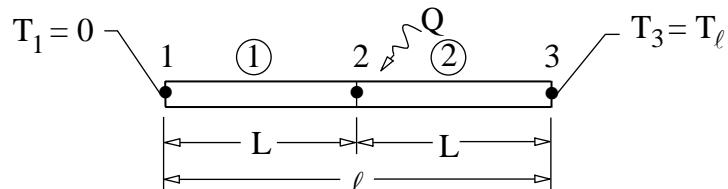
Conclusion of element equations:

$$[K_C] \{T\} = \{Q_C\} + \{Q_Q\}$$

$$\frac{kA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} = \begin{Bmatrix} -kA \frac{dT}{dx}(0) \\ kA \frac{dT}{dx}(L) \end{Bmatrix} + Q A L \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix}$$

METHOD OF WEIGHTED RESIDUALS

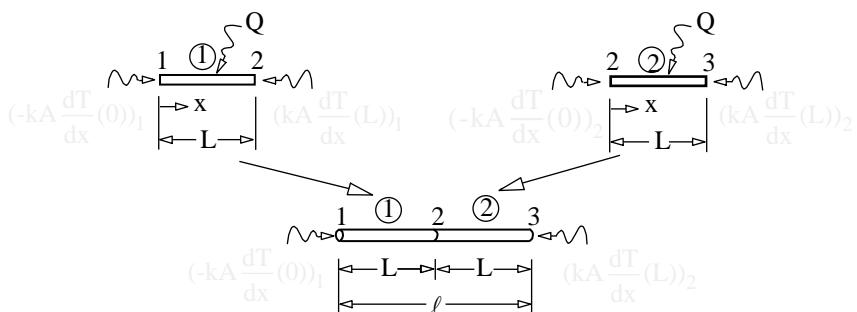
Example A rod divided into 2 elements with specified temp. at both ends as shown is subjected to internal heat generation Q . Compute the temperature at node ②.



Typical element eqs., e.g., for element no. ①:

$$\frac{kA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} = \begin{Bmatrix} (-kA \frac{dT}{dx}(0))_1 \\ (kA \frac{dT}{dx}(L))_1 \end{Bmatrix} + QAL \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix}$$

ASSEMBLING OF ELEMENT EQUATIONS



$$\frac{kA}{L} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1+1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \end{Bmatrix} = \begin{Bmatrix} \left(-kA \frac{dT}{dx}(0)\right)_1 \\ \left(kA \frac{dT}{dx}(L)\right)_1 + \left(-kA \frac{dT}{dx}(0)\right)_2 \\ \left(kA \frac{dT}{dx}(L)\right)_2 \end{Bmatrix} + QAL \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} + \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix}$$

SYSTEM OF EQUATIONS

Heat fluxes at node 2 must be continuous, and apply BC's:

$$\frac{kA}{L} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 2 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{Bmatrix} T_1 = 0 \\ T_2 = ? \\ T_3 = T_\ell \end{Bmatrix} = \begin{Bmatrix} (-kA \frac{dT}{dx}(0))_1 \\ 0 \\ (kA \frac{dT}{dx}(L))_2 \end{Bmatrix} + QAL \begin{Bmatrix} \frac{1}{2} \\ 1 \\ \frac{1}{2} \end{Bmatrix}$$

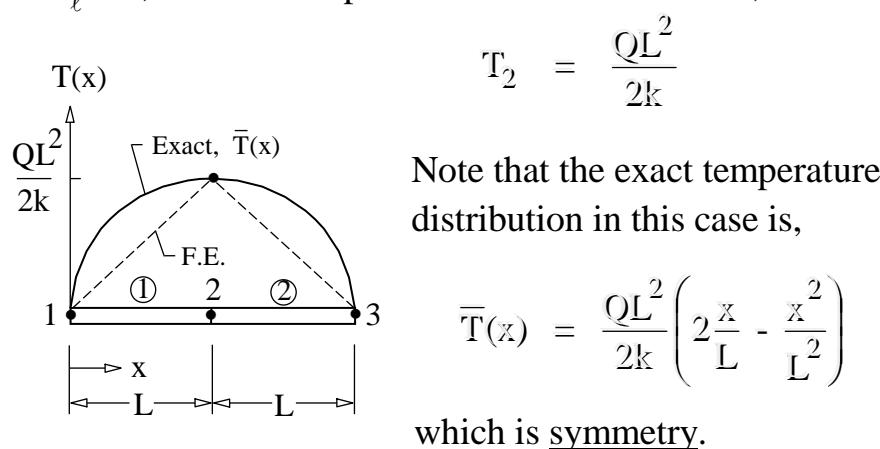
Then use second eq. to solve for temperature at node 2,

$$\begin{aligned} \frac{kA}{L} (0 + 2T_2 - T_\ell) &= 0 + QAL \\ T_2 &= \frac{QL^2}{2k} + \frac{T_\ell}{2} \end{aligned}$$

Note: Identical procedure can be used for more elements.

USE OF SOLUTION SYMMETRY

If $T_\ell = 0$, then the temperature at node 2 becomes,



USE OF SOLUTION SYMMETRY

Due to symmetry of solution, only half of the rod can be used in modeling. Here, if use 1 element,

$$\frac{kA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} = \begin{Bmatrix} (-kA \frac{dT}{dx}(0))_1 \\ (kA \frac{dT}{dx}(L))_1 \end{Bmatrix} + QAL \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix}$$

and apply the condition of no heat flow across node 2,
i.e.,

$$\left(kA \frac{dT}{dx}(L) \right)_1 = 0$$

Then the element eqs. become,

$$\frac{kA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} T_1 = 0 \\ T_2 = ? \end{Bmatrix} = \begin{Bmatrix} (-kA \frac{dT}{dx}(0))_1 \\ 0 \end{Bmatrix} + QAL \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix}$$

USE OF SOLUTION SYMMETRY

Then use the second eq. to solve for T_2 ,

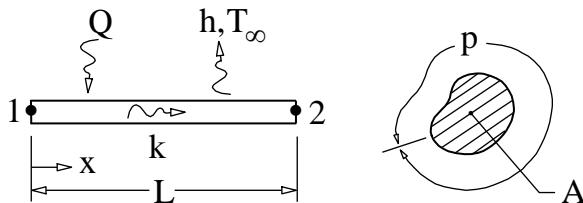
$$\begin{aligned} \frac{kA}{L} (0 + T_2) &= 0 + \frac{QAL}{2} \\ T_2 &= \frac{QL^2}{2k} \end{aligned}$$

Remark: For practical problems in 3-D, always take advantage of solution symmetry in order to reduce total number of elements and unknowns.

<u>Problem</u>	<u>Reduction factor</u>
1-D	2
2-D	4
3-D	8

ROD WITH SURFACE CONVECTION

If the rod has additional surface convection,



The corresponding differential equation is,

$$k A \frac{d^2 T}{dx^2} - h p (T - T_{\infty}) + QA = 0$$

where h = convection coefficient

p = perimeter

T_{∞} = surrounding medium temperature

ROD WITH SURFACE CONVECTION

Apply MWR,

$$\int_0^L W_i R dx = 0 \quad i = 1, 2$$

$$\int_0^L W_i (k A \frac{d^2 T}{dx^2} - h p (T - T_{\infty}) + QA) dx = 0$$

$$\begin{aligned} & \int_0^L W_i k A \frac{d^2 T}{dx^2} dx + \int_0^L W_i QA dx \\ & - \int_0^L W_i h p T dx + \int_0^L W_i h p T_{\infty} dx = 0 \end{aligned}$$

with $W_i = N_i$ and $T = \lfloor N \rfloor \{T\}$, the last two integral terms become,

ROD WITH SURFACE CONVECTION

$$\int_0^L W_i h p T dx \Rightarrow \int_0^L h p \{N\} [N] dx \{T\} \\ = [K_h] \{T\}$$

$$\int_0^L W_i h p T_\infty dx \Rightarrow \int_0^L h p T_\infty \{N\} dx \\ = \{Q_h\}$$

where $[K_h]$ = Convection matrix

$\{Q_h\}$ = Convection load vector

ROD WITH SURFACE CONVECTION

The finite element eqs. are in the form,

$$[K_c]\{T\} + [K_h]\{T\} = \{Q_c\} + \{Q_Q\} + \{Q_h\}$$

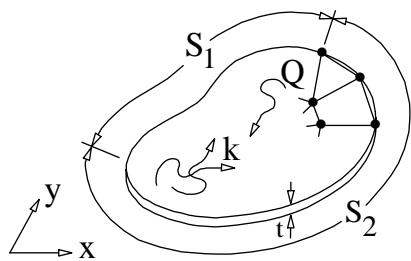
or, in detail,

$$\frac{kA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} + hpL \begin{bmatrix} \frac{1}{3} & \frac{1}{6} \\ \frac{1}{6} & \frac{1}{3} \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \end{Bmatrix} \\ = \begin{Bmatrix} -kA \frac{dT}{dx}(0) \\ kA \frac{dT}{dx}(L) \end{Bmatrix} + QAL \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix} + hpL T_\infty \begin{Bmatrix} \frac{1}{2} \\ \frac{1}{2} \end{Bmatrix}$$

which can be used in many applications, such as heat transfer in motor fins, etc.

TWO-DIMENSIONAL HEAT TRANSFER

Plate with Internal Heat Generation



Governing differential equation:

$$\frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial T}{\partial y} \right) = -Q$$

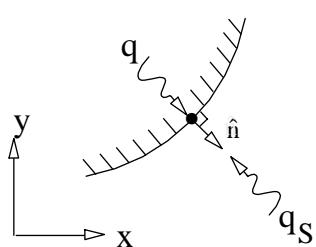
Boundary conditions:

- (1) Specified temperature along edge S_1 :

$$T(x, y) = T_1(x, y)$$

TWO-DIMENSIONAL HEAT TRANSFER

- (2) Specified surface heating along edge S_2 . From Fourier's law,



$q = -k \frac{\partial T}{\partial x} n_x - k \frac{\partial T}{\partial y} n_y$
where n_x and n_y are direction cosines of unit vector \hat{n} normal to the edge. For a specified heating q_s into edge, then

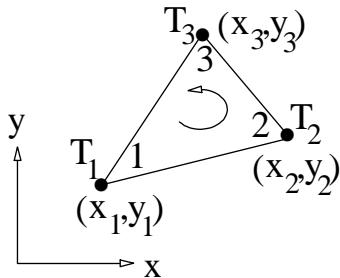
$$q_s = -q = k \frac{\partial T}{\partial x} n_x + k \frac{\partial T}{\partial y} n_y$$

This means, for insulated edge,

$$k \frac{\partial T}{\partial x} n_x + k \frac{\partial T}{\partial y} n_y = 0$$

TRIANGULAR ELEMENT

Element Interpolation Functions



Assume a flat temperature distribution over the element:

$T(x, y) = \alpha_1 + \alpha_2 x + \alpha_3 y$
where α_i , $i = 1, 2, 3$, are constants
to be determined from conditions
at nodes,

$$\text{At node 1: } T(x_1, y_1) = T_1 = \alpha_1 + \alpha_2 x_1 + \alpha_3 y_1$$

$$\text{At node 2: } T(x_2, y_2) = T_2 = \alpha_1 + \alpha_2 x_2 + \alpha_3 y_2$$

$$\text{At node 3: } T(x_3, y_3) = T_3 = \alpha_1 + \alpha_2 x_3 + \alpha_3 y_3$$

Solve for α_i , $i = 1, 2, 3$, and rearrange terms to get,

TRIANGULAR ELEMENT

$$T(x, y) = [N_1(x, y) \ N_2(x, y) \ N_3(x, y)] \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \end{Bmatrix} = [N] \{T\}$$

where $\{T\}$ is the vector of nodal temperatures, and $[N]$ is the element interpolation matrix defined by,

$$N_i(x, y) = \frac{1}{2A} (a_i + b_i x + c_i y) \quad i = 1, 2, 3$$

Here $A = \text{Area of triangle}$

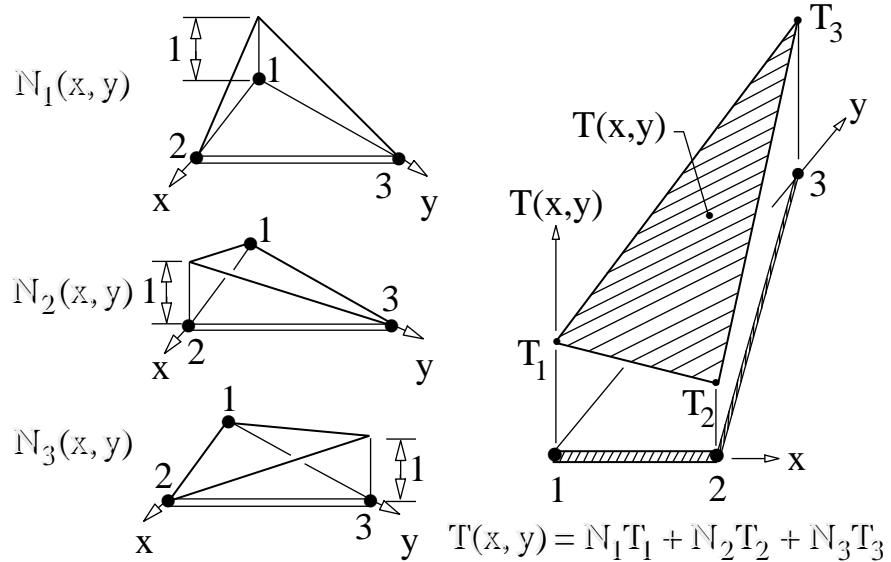
$$= \frac{1}{2} [x_2(y_3 - y_1) + x_1(y_2 - y_3) + x_3(y_1 - y_2)]$$

$$\text{and } a_1 = x_2 y_3 - x_3 y_2 \quad b_1 = y_2 - y_3 \quad c_1 = x_3 - x_2$$

$$a_2 = x_3 y_1 - x_1 y_3 \quad b_2 = y_3 - y_1 \quad c_2 = x_1 - x_3$$

$$a_3 = x_1 y_2 - x_2 y_1 \quad b_3 = y_1 - y_2 \quad c_3 = x_2 - x_1$$

TRIANGULAR ELEMENT INTERPOLATIONS



DERIVATION OF ELEMENT EQUATIONS

If \bar{T} is the exact temperature solution of the plate, then

$$\frac{\partial}{\partial x} \left(k \frac{\partial \bar{T}}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial \bar{T}}{\partial y} \right) + Q = 0$$

Since we use approximate T , then RHS of the above equation is not zero, but is equal to a Residual R , i.e.

$$\frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial T}{\partial y} \right) + Q = R$$

Apply MWR: $\int_{\Omega^{(e)}} W_i R d\Omega = 0$

where $\Omega^{(e)}$ is the element volume. Substitute,

$$\int_{\Omega^{(e)}} W_i \left(\frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial T}{\partial y} \right) + Q \right) d\Omega = 0$$

DERIVATION OF ELEMENT EQUATIONS

Expand,

$$\int_{\Omega^{(e)}} W_i \left(\frac{\partial}{\partial x} \left(k \frac{\partial \Gamma}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial \Gamma}{\partial y} \right) \right) d\Omega + \int_{\Omega^{(e)}} W_i Q d\Omega = 0$$

As in 1-D, we apply integration by parts to the first term,
Here is to use Gauss's theorem,

$$\int_{\Omega^{(e)}} u (\nabla \cdot \bar{V}) d\Omega = \int_{\Gamma^{(e)}} u (\bar{V} \cdot \hat{n}) d\Gamma - \int_{\Omega^{(e)}} (\nabla u \cdot \bar{V}) d\Omega$$

Let $u = W_i$

$$\begin{aligned} \nabla &= \frac{\partial}{\partial x} \hat{i} + \frac{\partial}{\partial y} \hat{j} \\ \bar{V} &= k \frac{\partial \Gamma}{\partial x} \hat{i} + k \frac{\partial \Gamma}{\partial y} \hat{j} \end{aligned} \quad (\nabla \cdot \bar{V}) = \left(\frac{\partial}{\partial x} \left(k \frac{\partial \Gamma}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial \Gamma}{\partial y} \right) \right)$$

DERIVATION OF ELEMENT EQUATIONS

Then the terms on RHS of Gauss's theorem which states as,

$$\int_{\Omega^{(e)}} u (\nabla \cdot \bar{V}) d\Omega = \int_{\Gamma^{(e)}} u (\bar{V} \cdot \hat{n}) d\Gamma - \int_{\Omega^{(e)}} (\nabla u \cdot \bar{V}) d\Omega$$

using $\hat{n} = n_x \hat{i} + n_y \hat{j}$, are:

$$\begin{aligned} \bar{V} \cdot \hat{n} &= k \frac{\partial \Gamma}{\partial x} n_x + k \frac{\partial \Gamma}{\partial y} n_y \\ u(\bar{V} \cdot \hat{n}) &= W_i \left(k \frac{\partial \Gamma}{\partial x} n_x + k \frac{\partial \Gamma}{\partial y} n_y \right) \\ \nabla u &= \frac{\partial W_i}{\partial x} \hat{i} + \frac{\partial W_i}{\partial y} \hat{j} \\ \nabla u \cdot \bar{V} &= \frac{\partial W_i}{\partial x} k \frac{\partial \Gamma}{\partial x} + \frac{\partial W_i}{\partial y} k \frac{\partial \Gamma}{\partial y} \end{aligned}$$

DERIVATION OF ELEMENT EQUATIONS

Thus the element equations with $W_i = N_i$ are,

$$\int_{\Gamma^{(e)}} N_i \left(k \frac{\partial T}{\partial x} n_x + k \frac{\partial T}{\partial y} n_y \right) d\Gamma - \int_{\Omega^{(e)}} \left(\frac{\partial N_i}{\partial x} k \frac{\partial T}{\partial x} + \frac{\partial N_i}{\partial y} k \frac{\partial T}{\partial y} \right) d\Omega \\ + \int_{\Omega^{(e)}} N_i Q d\Omega = 0 \quad i = 1, 2, 3$$

Imposing the boundary conditions along edges, the element equations can be written in matrix form (total of 3 eqs.) as,

$$\int_{\Omega^{(e)}} \left(\begin{Bmatrix} \frac{\partial N}{\partial x} \\ \frac{\partial N}{\partial y} \end{Bmatrix} k \begin{Bmatrix} \frac{\partial T}{\partial x} \\ \frac{\partial T}{\partial y} \end{Bmatrix} \right) d\Omega \\ = \int_{\Gamma^{(e)}} \{N\} \left(k \frac{\partial T}{\partial x} n_x + k \frac{\partial T}{\partial y} n_y \right) d\Gamma + \int_{\Omega^{(e)}} \{N\} Q d\Omega + \int_{\Gamma^{(e)}} \{N\} q_s d\Gamma$$

DERIVATION OF ELEMENT EQUATIONS

Since we assume element temperature distribution,

$$T = T(x, y) = \begin{bmatrix} N \end{bmatrix} \{T\}$$

$$\frac{\partial T}{\partial x} = \begin{bmatrix} \frac{\partial N}{\partial x} \end{bmatrix} \{T\}$$

and

$$\frac{\partial T}{\partial y} = \begin{bmatrix} \frac{\partial N}{\partial y} \end{bmatrix} \{T\}$$

Substitute to get the final form of element equations,

DERIVATION OF ELEMENT EQUATIONS

$$\begin{aligned}
 & \underbrace{\int_{\Omega^{(e)}} \left(\begin{bmatrix} \frac{\partial N}{\partial x} \\ (3x1) \end{bmatrix}_k \begin{bmatrix} \frac{\partial N}{\partial x} \\ (1x3) \end{bmatrix} + \begin{bmatrix} \frac{\partial N}{\partial y} \\ (3x1) \end{bmatrix}_k \begin{bmatrix} \frac{\partial N}{\partial y} \\ (1x3) \end{bmatrix} \right) d\Omega}_{[K_e] \quad (3x3)} \{T\}_{(3x1)} \\
 & = \underbrace{\int_{\Gamma^{(e)}} \{N\}_{(3x1)} \left(k \frac{\partial T}{\partial x} n_x + k \frac{\partial T}{\partial y} n_y \right) d\Gamma}_{\{Q_c\}_{(3x1)}} + \underbrace{\int_{\Omega^{(e)}} \{N\}_{(3x1)} Q d\Omega}_{\{Q_Q\}_{(3x1)}} + \underbrace{\int_{S_2} \{N\}_{(3x1)} q_s d\Gamma}_{\{Q_q\}_{(3x1)}}
 \end{aligned}$$

Or, in short (details given in text):

$$[K_e]\{T\} = \{Q_c\} + \{Q_Q\} + \{Q_q\}$$

FINITE ELEMENT MATRICES

Element matrices for triangle can be derived in closed-form. Since,

$$N_i = \frac{1}{2A} (a_i + b_i x + c_i y) \quad i = 1, 2, 3$$

then $\frac{\partial N_i}{\partial x} = \frac{b_i}{2A}$ and $\frac{\partial N_i}{\partial y} = \frac{c_i}{2A}$

Conduction matrix Coefficients in the matrix are,

$$K_{ij} = \int_A k \left(\frac{\partial N_i}{\partial x} \frac{\partial N_j}{\partial x} + \frac{\partial N_i}{\partial y} \frac{\partial N_j}{\partial y} \right) t dx dy \quad i, j = 1, 2, 3$$

For constant thermal conductivity k and thickness t , then

$$K_{ij} = k t \int_A \left(\frac{b_i}{2A} \frac{b_j}{2A} + \frac{c_i}{2A} \frac{c_j}{2A} \right) dx dy$$

FINITE ELEMENT MATRICES

Or, $K_{ij} = \frac{k t}{4A} (b_i b_j + c_i c_j) \quad i, j = 1, 2, 3$

$$[K_c] = k A t [B]^T [B] \text{ where } [B] = \frac{1}{2A} \begin{bmatrix} b_1 & b_2 & b_3 \\ c_1 & c_2 & c_3 \end{bmatrix}$$

Load vector due to heat generation Coefficients are,

$$Q_i = \int_A N_i Q t dx dy \quad i = 1, 2, 3$$

For constant Q and t over element, integrate to get,

$$Q_i = \frac{Q A t}{3}$$

FINITE ELEMENT MATRICES

Or, in matrix form: $\{Q_Q\} = \frac{Q A t}{3} \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix}$

Load vector due to edge heating

$$\{Q_q\} = \int_{s_2} \{N\} q_s d\Gamma \quad \text{E.g.,} \quad \{Q_q\} = \frac{q_s t \ell_{12}}{2} \begin{Bmatrix} 1 \\ 1 \\ 0 \end{Bmatrix}$$

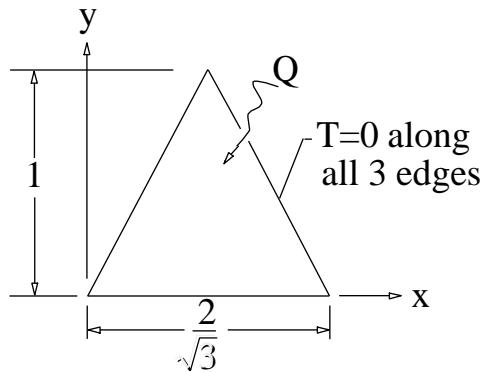
having nodes 1 and 2 on the edge.

Load vector for conduction

$$\{Q_c\} = \int_{\Gamma^{(e)}} \{N\} \left(k \frac{\partial T}{\partial x} n_x + k \frac{\partial T}{\partial y} n_y \right) d\Gamma$$

TWO-DIMENSIONAL HEAT TRANSFER

Example Determine temperature distribution in a triangular plate with internal heat generation using 3 finite elements.

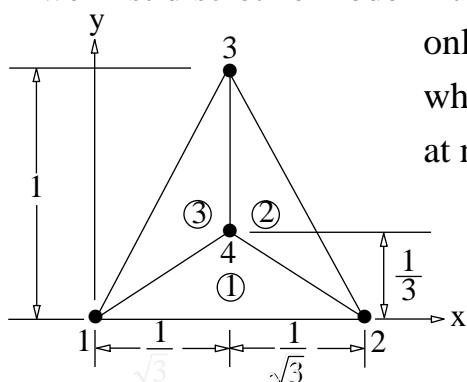


Note: This example has exact temperature solution as,

$$\bar{T}(x, y) = \frac{Q}{4k} (y - 2 + \sqrt{3}x)(y - \sqrt{3}x)y$$

TWO-DIMENSIONAL HEAT TRANSFER

we first discretize model into 3 elements. Here we have only one unknown at node 4 where as temperatures are zero at nodes 1, 2, and 3.



Typical element eqs. are:

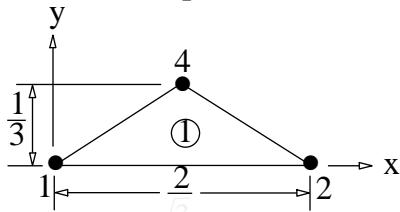
$$[K_c]\{T\} = \{Q_c\} + \{Q_Q\}$$

e.g., for element no. 1,

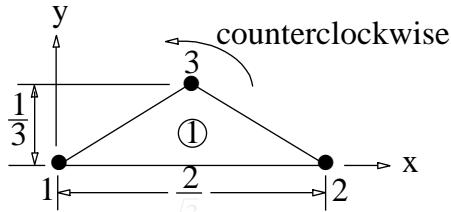
$$\begin{bmatrix} k_{11} & k_{12} & k_{13} \\ k_{21} & k_{22} & k_{23} \\ k_{31} & k_{32} & k_{33} \end{bmatrix} \begin{bmatrix} T_1 \\ T_2 \\ T_3 \end{bmatrix} = \begin{bmatrix} Q_{c1} \\ Q_{c2} \\ Q_{c3} \end{bmatrix} + \frac{QAt}{3} \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix}$$

TWO-DIMENSIONAL HEAT TRANSFER

Computation of element matrices:



Actual node numbers



Node numbers in derivation

$$\begin{aligned} \text{Here, } x_1 &= 0 & y_1 &= 0 & b_1 &= -\frac{1}{3} & c_1 &= -\frac{1}{\sqrt{3}} \\ x_2 &= \frac{2}{\sqrt{3}} & y_2 &= 0 & b_2 &= \frac{1}{3} & c_2 &= -\frac{1}{\sqrt{3}} \\ x_3 &= \frac{1}{\sqrt{3}} & y_3 &= \frac{1}{3} & b_3 &= 0 & c_3 &= \frac{2}{\sqrt{3}} \end{aligned}$$

with the element area $A = 1/3\sqrt{3}$.

TWO-DIMENSIONAL HEAT TRANSFER

With these coefficients b_i and c_i , $i = 1, 2, 3$, element matrices can be determined. For example the coefficient K_{23} in the conduction matrix is,

$$\begin{aligned} K_{23} &= \frac{k t}{4A} (b_2 b_3 + c_2 c_3) \\ &= \frac{k t}{4 \left(\frac{1}{3\sqrt{3}} \right)} \left(\left(\frac{1}{3} \right)(0) + \left(-\frac{1}{\sqrt{3}} \right) \left(\frac{2}{\sqrt{3}} \right) \right) = k t \left(-\frac{3}{2\sqrt{3}} \right) \end{aligned}$$

Thus, element equations for element no. ① are,

$$\frac{k t}{2\sqrt{3}} \begin{bmatrix} 2 & 1 & -3 \\ 1 & 2 & -3 \\ -3 & -3 & 6 \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \\ T_4 \end{Bmatrix} = \begin{Bmatrix} (Q_{c_1})_1 \\ (Q_{c_2})_1 \\ (Q_{c_4})_1 \end{Bmatrix} + \frac{Q t}{9\sqrt{3}} \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix}$$

TWO-DIMENSIONAL HEAT TRANSFER

Similary, element eqs. for element no. ② are

$$\frac{k t}{2\sqrt{3}} \begin{bmatrix} 2 & 1 & -3 \\ 1 & 2 & -3 \\ -3 & -3 & 6 \end{bmatrix} \begin{Bmatrix} T_2 \\ T_3 \\ T_4 \end{Bmatrix} = \begin{Bmatrix} (Q_{c_2})_2 \\ (Q_{c_3})_2 \\ (Q_{c_4})_2 \end{Bmatrix} + \frac{Q t}{9\sqrt{3}} \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix}$$

and for element no. ③ are,

$$\frac{k t}{2\sqrt{3}} \begin{bmatrix} 2 & 1 & -3 \\ 1 & 2 & -3 \\ -3 & -3 & 6 \end{bmatrix} \begin{Bmatrix} T_3 \\ T_1 \\ T_4 \end{Bmatrix} = \begin{Bmatrix} (Q_{c_3})_3 \\ (Q_{c_1})_3 \\ (Q_{c_4})_3 \end{Bmatrix} + \frac{Q t}{9\sqrt{3}} \begin{Bmatrix} 1 \\ 1 \\ 1 \end{Bmatrix}$$

Then assemble element equations. from these 3 elements to get,

TWO-DIMENSIONAL HEAT TRANSFER

$$\frac{k t}{2\sqrt{3}} \begin{bmatrix} 2+2 & 1 & 1 & -3-3 \\ 1 & 2+2 & 1 & -3-3 \\ 1 & 1 & 2+2 & -3-3 \\ -3-3 & -3-3 & -3-3 & 6+6+6 \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \end{Bmatrix} = \begin{cases} \frac{(Q_{c_1})_1 + (Q_{c_1})_3}{1+1} \\ \frac{(Q_{c_2})_1 + (Q_{c_2})_2}{1+1} \\ \frac{(Q_{c_3})_2 + (Q_{c_3})_3}{1+1} \\ \frac{(Q_{c_4})_1 + (Q_{c_4})_2 + (Q_{c_4})_3}{1+1+1} \end{cases} + \frac{Q t}{9\sqrt{3}} \begin{Bmatrix} 1+1 \\ 1+1 \\ 1+1 \\ 1+1+1 \end{Bmatrix}$$

TWO-DIMENSIONAL HEAT TRANSFER

Apply boundary conditions of $T_1 = T_2 = T_3 = 0$, the system eqs. become,

$$\frac{k t}{2\sqrt{3}} \begin{bmatrix} 4 & 1 & 1 & -6 \\ 1 & 4 & 1 & -6 \\ 1 & 1 & 4 & -6 \\ -6 & -6 & -6 & 18 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 0 \\ T_4 \end{bmatrix} = \begin{bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ 0 \end{bmatrix} + \frac{Q t}{9\sqrt{3}} \begin{bmatrix} 2 \\ 2 \\ 2 \\ 3 \end{bmatrix}$$

Then use the last equation to solve for temperature at node 4,

$$\frac{k t}{2\sqrt{3}} (0 + 0 + 0 + 18 T_4) = 0 + \frac{Q t}{9\sqrt{3}} (3)$$

$$T_4 = \frac{1}{27} \frac{Q}{k}$$

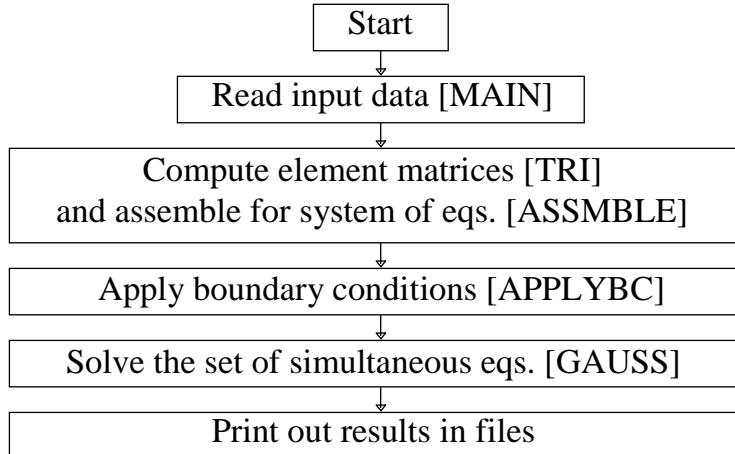
and then can use the first 3 equations to solve for heat fluxes at nodes,

$$Q_1 = Q_2 = Q_3 = -\frac{6 k t}{2\sqrt{3}} T_4 - \frac{2 Q t}{9\sqrt{3}} = -\frac{Q t}{3\sqrt{3}}$$

COMPUTER PROGRAMMING

- Same procedure described in example can be used in the development of finite element computer program directly.
- Listing of sample finite element program provided in following pages (as short as only 6 A-4 pages).
- Program is applicable to arbitrary geometry.
- Program can be modified to solve other problems.
- By understanding this short program -- won't be afraid of "black box" anymore.

COMPUTER PROGRAMMING FLOW-CHART



Note: Names in the square brackets [] represent the subroutine names in the finite element program illustrated.

FINITE ELEMENT PROGRAM LISTING 230

Example in FORTRAN

```

PROGRAM FINITE
C A FINITE ELEMENT COMPUTER PROGRAM FOR SOLVING PARTIAL
C DIFFERENTIAL EQUATION IN THE FORM OF POISSON'S EQUATION
C FOR TWO-DIMENSIONAL STEADY-STATE HEAT CONDUCTION WITH
C INTERNAL HEAT GENERATION.
      DR. PRAMOD DEOGENGI
      FACULTY OF CIVIL ENGINEERING
      CHULALONGKORN UNIVERSITY

C THE VALUES DECLARED IN THE PARAMETER STATEMENT BELOW SHOULD
C BE ADJUSTED ACCORDING TO THE SIZE OF THE PROBLEM AND TYPES
C OF COMPUTERS:
      NMAX = MAXIMUM NUMBER OF NODES IN THE MODEL
      NELEM = MAXIMUM NUMBER OF ELEMENTS IN THE MODEL

PARAMETER (NOD=24,150, MNELEM=300)

REAL A(1,1), B(1,1), C(1,1)
REAL X(1,1), Y(1,1), TEMP(MNELEM), TEXT(20)
DIMENSION SYX(MNELEM,MNELEM), SYQ(MNELEM), QELM(MNELEM)
CHARACTER*20 NAME1, NAME2

INTEGER INTMAX(MNELEM,3), IBC(MNELEM)

10 WRITE(6,20)
20 FORMAT(//, * PLEASE ENTER THE INPUT FILE NAME: ')
READ(5, *, IHR=10) NAME1
OPEN(UNIT=11, FILE=NAME1, STATUS='OLD', IHR=10)

READ TITLE OF COMPUTATION:
READ(1,*), NLINES
DO 100 ILINE=1,NLINES
READ(1,1), TEXT
100 FORMAT(1A)
100 CONTINUE

READ INPUT DATA:
READ(1,1), TEXT
READ(1,*), NPOINT, NELEM
IF(NPOINT.GT.MNELEM) WRITE(6,110) NPOINT
110 FORMAT(//, * PLEASE INCREASE THE PARAMETER MNELEM TO ', IS')
IF(NPOINT.LT.MNELEM) WRITE(6,120) NPOINT
120 FORMAT(//, * PLEASE INCREASE THE PARAMETER MNELEM TO ', IS')
IF(NPOINT.EQ.MNELEM) STOP
READ(7,1) TEXT
READ(7,*), TX, THICK
READ(7,*), RHO, K
DO 130 IP=1,NPOINT
READ(7,*), IBC(I,1), (COORH(I,K), K=1,2), TEMP(I)
130 FORMAT(1A, 2I5, 2E15.15) IS
135 FORMAT(//, * NODE NO., IS, ' IN DATA FILE IS MISSING')
140 IF(I.NE.IP) STOP
140 CONTINUE
140 IQ = 0
READ(7,1) TEXT
DO 140 IE=1,NELEM

```

FINITE ELEMENT PROGRAM LISTING

```

SUBROUTINE APPLBC(IEBCIN, IBC, TEMP, SYRK, SYSQ, MXPOI)
C
C APPLY TEMPERATURE BOUNDARY CONDITIONS WITH CONDITION CODES OF:
C   1 = FREE TO RANGE (TO BE COMPUTED)
C   2 = FIXED AS SPECIFIED
C
C IMPLICIT REAL*8 (A=H,O=2)
DIMENSION SYR(MXPOI,MXPOI), SYSQ(MXPOI), TEMP(MXPOI)
C
INTEGER IBC(IPXPOI)
C
DO 100 IBC=1,MXPOI
IF(IEBC(IPXPOI,IBC,0)) GO TO 100
C
DO 200 IBC=1,MXPOI
IF(IEBC(IPXPOI,IBC,0)) GO TO 200
SYRK(IBC) = SYRK(IBC) - TEMP(IBC)*TEMP(IBC)
SYSQ(IBC) = 0.
200 CONTINUE
C
DO 300 IBC=1,MXPOI
SYRK(IBC,IBC) = 0.
300 CONTINUE
SYRK(IBC,IBC) = 1.
SYSQ(IBC) = TEMP(IBC)
C
100 CONTINUE
C
RETURN
END
C
C-----
C SUBROUTINE ASSEMBLE( IE, INTMAT, AMK, QQ, SYRK, SYSQ,
      *           MXPOI, MXEL )
C
C ASSEMBLE ELEMENT EQUATIONS INTO SYSTEM EQUATIONS
C
IMPLICIT REAL*8 (A=H,O=2)
DIMENSION AMK(3,3), QQ(3)
DIMENSION SYR(MXPOI,MXPOI), SYSQ(MXPOI)
C
INTEGER INTMAT(MXEL,E)
C
NNODES = 3
C
DO 100 IBC=1,MXPOI
DO 200 IBC=1,MXPOI
  IRW = INTMAT(IE,IBC)
  DO 300 IC=1,MXPOI
    SYRK(IRW,IC) = SYRK(IRW,IC) + AMK(IC,IC)
  300 CONTINUE
  SYRK(IRW,IRW) = SYRK(IRW,IRW) + QQ(IE)
100 CONTINUE
C
  RETURN
END
C
C-----
C SUBROUTINE GAUSS(N, A, B, X, MXPOI)
IMPLICIT REAL*8 (A=H,O=2)
DIMENSION A(MXPOI,MXPOI), B(MXPOI), X(MXPOI)
C
PERFORM SCALING:
CALL SCALE(N, A, B, MXPOI)

```

```

C FORWARD ELIMINATION:
C PERFORM ACCORDING TO ORDER OF 'PRIME' FROM 1 TO N-1:
DO 100 IIP=1,N
C
C PERFORM PARTIAL PIVOTING:
CALL PIVOTN( A, B, MXPOI, IP )
C
C LOOP OVER EACH EQUATION STARTING FROM THE ONE THAT CORRESPONDS
C WITH THE ORDER OF 'PRIME' PLUS ONE:
DO 200 IIP=IP+1,N
  RATIO = A(IIP,IP)/A(IP,IP)
C
C COMPUTE NEW COEFFICIENTS OF THE EQUATION CONSIDERED:
DO 300 IC=IP+1,N
    A(IC,IIP) = A(IC,IC) - RATIO*A(IP,IC)
  300 CONTINUE
  B(IIP) = B(IIP) - RATIO*B(IP)
200 CONTINUE
C
C SET COEFFICIENTS ON LOWER LEFT PORTION TO ZERO:
DO 400 IC=IP+1,N
    A(IC,IIP) = A(IC,IC)
  400 CONTINUE
100 CONTINUE
C
C BACK SUBSTITUTION:
C COMPUTE SOLUTION OF THE LAST EQUATION:
X(N) = B(N)/A(N,N)
C
C THEN COMPUTE SOLUTIONS FROM EQUATION N-1 TO 1:
DO 500 IIP=N-1,1,-1
  SUM = 0.
  DO 600 IC=IP+1,N
    SUM = SUM + A(IC,IIP)*X(IC)
  600 CONTINUE
  X(IIP) = (B(IIP) - SUM)/A(IP,IIP)
500 CONTINUE
C
C-----
```

```

SUBROUTINE PIVOTN( A, B, MXPOI, IP )
IMPLICIT REAL*8 (A=H,O=2)
DIMENSION A(MXPOI,MXPOI), B(MXPOI)
C
C PERFORM PARTIAL PIVOTING:

```

```

JP = IP
BIG = ABS(A(IP,IP))
DO 10 I=IP+1,N
  AMAX = ABS(A(I,IP))
  IF(AMAX.GT.BIG) THEN
    BIG = AMAX
    JP = I
  ENDIF
10 CONTINUE

```

FINITE ELEMENT PROGRAM LISTING²³²

```

IF(AMAX.GT.BIG) THEN
  BIG = AMAX
  JP = I
ENDIF
C
DO 20 J=JP+1,N
  IF(AMAX.NE.BIG) THEN
    DO 30 I=1,N
      A(IP,J) = A(IP,J)/BIG
      A(JP,J) = A(IP,J)
      A(IP,I) = DUMY
20 CONTINUE
  ENDIF
  DO 40 I=1,N
    B(IP,I) = B(IP,I)/BIG
    B(IP,I) = DUMY
  40 CONTINUE
  RETURN
END
C
C-----
```

```

SUBROUTINE SCALE(N, A, B, MXPOI)
IMPLICIT REAL*8 (A=H,O=2)
DIMENSION A(MXPOI,MXPOI), B(MXPOI)
C
C PERFORM SCALING:

```

```

DO 10 IBC=1,N
  BIG = ABS(A(1,IBC))
  DO 20 IC=1,N
    AMAX = ABS(A(IC,IBC))
    IF(AMAX.GT.BIG) BIG = AMAX
20 CONTINUE
  A(1,IBC) = A(1,IBC)/BIG
  B(1,IBC) = B(1,IBC)/BIG
10 CONTINUE
C
  RETURN
END
C
C-----
```

```

SUBROUTINE TH(NELEM, INTMAT, COORD, TR, QLKE, THICK,
      *           SYRK, MXPOI, MXEL )
C
C ESTABLISH ALL ELEMENT MATRICES AND ASSEMBLE THEM TO FORM
C UP SYSTEM EQUATIONS:

```

```

IMPLICIT REAL*8 (A=H,O=2)
DIMENSION COORD(3,1,2), SYRK(MXPOI,MXPOI), SYSQ(MXPOI)
DIMENSION INTMAT(MXEL,E), MXEL
DIMENSION AMK(3,3), QQ(3), B(2,3), BT(3,2)
C
INTEGER INTMAT(MXEL,E)
C
LOOP OVER THE NUMBER OF ELEMENTS:
DO 500 IEL=1,NELEM
C
C FIND ELEMENT LOCAL COORDINATES:
C
IEL = INTMAT(IE,1)
IE1 = INTMAT(IE,2)
IE2 = INTMAT(IE,3)
C
X01 = COORD(1,IEL)
X02 = COORD(2,IEL)
X03 = COORD(3,IEL)

```

```

Y01 = COORD(1,IEL)
Y02 = COORD(2,IEL)
Y03 = COORD(3,IEL)
AREA = 0.5*(X02*(Y03-Y01)) + X03*(Y01-Y02)
BT = AREA/(3.0*SQRT(1.0+X01*X01+X02*X02+X03*X03))
5 FORMAT(1X, 'ELEMENT NO.', I5,
      *       ' HAS INTEGRAL OR ZERO AREA',
      *       ' OR NO. OF CURENT FLOW AREAS FOR NODAL COORDINATES',
      *       ' AND ELEMENT NODAL CONNECTIONS ---')
IF(AREA.LE.0.) STOP
```

```

B1 = Y02 - Y03
B2 = Y01 - Y03
B3 = Y01 - Y02
C1 = X03 - X02
C2 = X03 - X01
C3 = X02 - X01

```

```

DO 10 I=1,2
  DO 20 J=1,3
    BT(I,J) = BT(I,J)
10 CONTINUE

```

```

B1(1,1) = B1
B1(1,2) = B2
B1(1,3) = B3
B2(1,2) = C1
B2(1,3) = C2
B3(1,3) = C1

```

```

DO 30 I=2,3
  DO 40 J=1,2
    BT(I,J) = B(I,J)/(2.*AREA)
    BT(I,J) = BT(I,J)
30 CONTINUE
20 CONTINUE

```

```

ELEMENT CONDUCTION MATRIX:
DO 100 I=1,3
  DO 200 J=1,3
    AMK(I,J) = AMK(I,J) + BT(I,J)*BT(K,J)
100 CONTINUE
AMK(1,1) = TR*AREA*THICK*AMK(1,1)
100 CONTINUE

```

```

ELEMENT LOAD VECTOR DUE TO INTERNAL HEAT GENERATION:
PAC = QLKE(IE)*AREA*THICK/3.
DO 200 I=1,3
  DO 300 J=1,2
    PAC(I,J) = PAC(I,J) + AMK(I,J)*BT(I,J)
300 CONTINUE
200 CONTINUE

```

```

ASSEMBLE THESE ELEMENT MATRICES TO FORM SYSTEM EQUATIONS:
CALL ASSEMBLE( IE, INTMAT, AMK, QQ, SYRK, SYSQ,
      *           MXPOI, MXEL )

```

```

500 CONTINUE
RETURN
END
C

```

TYPICAL INPUT & OUTPUT DATA

Input data

```

2
TRIANGULAR PLATE WITH INTERNAL HEAT GENERATION.
CRUDE MESH WITH 4 NODES AND 3 ELEMENTS.
NPOIN    NELEM
        4      3
TK      THICK
1.      .1
NODAL BOUNDARY CONDITIONS AND COORDINATES [4]:
1       1       0.00000   0.00000   0.
2       1       1.15470   0.00000   0.
3       1       0.57735   1.00000   0.
4       0       0.57735   0.33333   0.
ELEMENT NODAL CONNECTIONS AND HEAT GEN. [3]:
1       1       2       4       1.
2       2       3       4       1.
3       3       1       4       1.

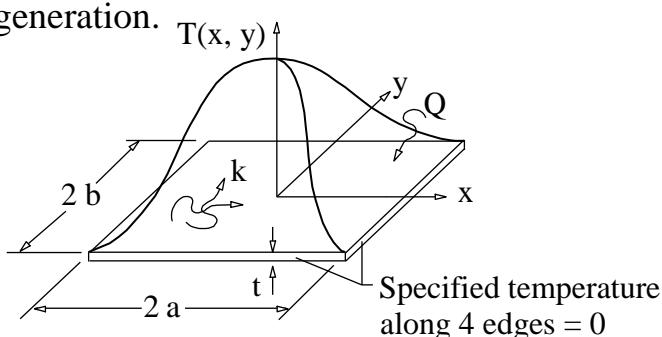
```

Output result

NODE	TEMPERATURE
1	.000000E+00
2	.000000E+00
3	.000000E+00
4	.370370E-01

RECTANGULAR ELEMENT

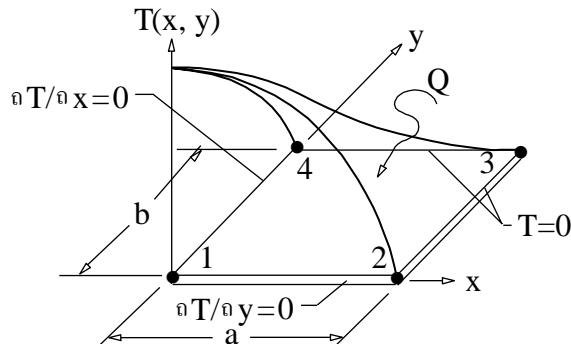
Example Two-dimensional steady-state heat conduction in a rectangular plate with internal heat generation.



Due to symmetry, only a quarter of the plate can be modeled. Use one 4-node rectangular element to compute temperature at the plate center ($x=y=0$).

RECTANGULAR ELEMENT

By using only one rectangular element for quarter of the plate, some edge conditions must be provided,



Here

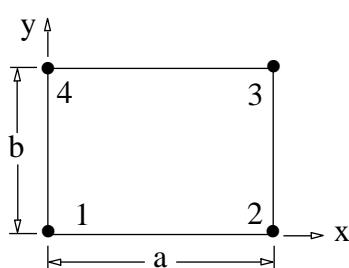
$$T_2 = T_3 = T_4 = 0$$

Want

$$T_1 = ?$$

RECTANGULAR ELEMENT

Element interpolation function:



Assume element temperature,

$$T(x, y) = \alpha_1 + \alpha_2 x + \alpha_3 y + \alpha_4 xy$$

which is in form of bilinear distribution where α_i , $i = 1, 4$ are constants that can be determined from the conditions,

$$T(0,0) = T_1 = \alpha_1 + 0 + 0 + 0$$

$$T(a,0) = T_2 = \alpha_1 + \alpha_2 a + 0 + 0$$

$$T(a,b) = T_3 = \alpha_1 + \alpha_2 a + \alpha_3 b + \alpha_4 ab$$

$$T(0,b) = T_4 = \alpha_1 + 0 + \alpha_3 b + 0$$

RECTANGULAR ELEMENT

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Solve for $\alpha_i, i=1, 4$ to get,

$$\begin{aligned}\alpha_1 &= T_1 & \alpha_3 &= \frac{T_4 - T_1}{b} \\ \alpha_2 &= \frac{T_2 - T_1}{a} & \alpha_4 &= \frac{T_1 + T_2 + T_3 - T_4}{ab}\end{aligned}$$

Then, substitute back and rearrange terms to get element temperature distribution in the form,

$$\begin{aligned}T &= \begin{bmatrix} (1 - \frac{x}{a})(1 - \frac{y}{b}) & \frac{x}{a}(1 - \frac{y}{b}) & \frac{x}{a}\frac{y}{b} & \frac{y}{b}(1 - \frac{x}{a}) \\ \underbrace{N_1}_{(1x4)} & \underbrace{N_2}_{(4x1)} & \underbrace{N_3}_{(1x4)} & \underbrace{N_4}_{(4x1)} \end{bmatrix} \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \end{Bmatrix} \\ &= [N(x, y)] \{T\}\end{aligned}$$

where $[N(x, y)]$ is the element interpolation function matrix.

PLATE WITH INTERNAL HEAT GEN.

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Since the corresponding FE eqs. for this problem are,

$$[K_c]\{T\} = \{Q_c\} + \{Q_Q\}$$

where $[K_c]$ is the conduction matrix,

$$[K_c] = \int_{\Omega^{(e)}} k \left(\left\{ \frac{\partial N}{\partial x} \right\} \left[\frac{\partial N}{\partial x} \right] + \left\{ \frac{\partial N}{\partial y} \right\} \left[\frac{\partial N}{\partial y} \right] \right) d\Omega$$

$$\text{Here, } \left[\frac{\partial N}{\partial x} \right] = \left[-\frac{1}{a}(1 - \frac{y}{b}) \quad \frac{1}{a}(1 - \frac{y}{b}) \quad \frac{1}{a}\frac{y}{b} \quad -\frac{1}{a}\frac{y}{b} \right]$$

$$\left[\frac{\partial N}{\partial y} \right] = \left[-\frac{1}{b}(1 - \frac{x}{a}) \quad -\frac{1}{b}\frac{x}{a} \quad \frac{1}{b}\frac{x}{a} \quad \frac{1}{b}(1 - \frac{x}{a}) \right]$$

and $d\Omega = t dx dy$. Perform integration with the limits $x = 0 \rightarrow a$ and $y = 0 \rightarrow b$ to obtain,

FINITE ELEMENT EQUATIONS

Conduction matrix,

$$[K_c] = k t \begin{bmatrix} \frac{1}{3} \left(\frac{a+b}{b-a} \right) & \frac{1}{6} \left(\frac{a-2b}{b-a} \right) & \frac{1}{6} \left(-\frac{a-b}{b-a} \right) & \frac{1}{6} \left(-\frac{2a+b}{b-a} \right) \\ \frac{1}{6} \left(\frac{a+b}{b-a} \right) & \frac{1}{3} \left(-\frac{2a+b}{b-a} \right) & \frac{1}{6} \left(-\frac{a-b}{b-a} \right) & \frac{1}{6} \left(-\frac{a+b}{b-a} \right) \\ \frac{1}{6} \left(\frac{a+b}{b-a} \right) & \frac{1}{6} \left(-\frac{a-b}{b-a} \right) & \frac{1}{3} \left(\frac{a+b}{b-a} \right) & \frac{1}{6} \left(\frac{a-2b}{b-a} \right) \\ \text{sym} & & \frac{1}{6} \left(\frac{a+b}{b-a} \right) & \frac{1}{3} \left(\frac{a+b}{b-a} \right) \end{bmatrix}$$

Also, the load vector from internal heat generation,

$$\{Q_Q\} = \int_{\Omega^{(e)}} Q \{N\} d\Omega = Q a b t \begin{Bmatrix} \frac{1}{4} \\ \frac{1}{4} \\ \frac{1}{4} \\ \frac{1}{4} \\ \frac{1}{4} \\ \frac{1}{4} \end{Bmatrix}$$

FINITE ELEMENT EQUATIONS

and the conduction vector,

$$\{Q_c\} = \int_{\Gamma^{(e)}} \{N\} k \left(\frac{\partial T}{\partial x} n_x + \frac{\partial T}{\partial y} n_y \right) d\Gamma = \begin{Bmatrix} Q_{c_1} \\ Q_{c_2} \\ Q_{c_3} \\ Q_{c_4} \end{Bmatrix}$$

Therefore, the final element eqs. can be written as,

$$\begin{bmatrix} K_{11} & K_{12} & K_{13} & K_{14} \\ K_{21} & K_{22} & K_{23} & K_{24} \\ K_{31} & K_{32} & K_{33} & K_{34} \\ \text{Sym} & & K_{43} & K_{44} \end{bmatrix} \begin{Bmatrix} T_1 = ? \\ T_2 = 0 \\ T_3 = 0 \\ T_4 = 0 \end{Bmatrix} = \begin{Bmatrix} Q_{c_1} \\ Q_{c_2} \\ Q_{c_3} \\ Q_{c_4} \end{Bmatrix} + Q a b t \begin{Bmatrix} \frac{1}{4} \\ \frac{1}{4} \\ \frac{1}{4} \\ \frac{1}{4} \end{Bmatrix}$$

Then use the first eq. to solve for T_1 ,

PLATE WITH INTERNAL HEAT GEN.

$$K_{11}T_1 + 0 + 0 + 0 = Q_{c_1} + \frac{Qabt}{4}$$

$$\frac{k}{3}t\left(\frac{a}{b} + \frac{b}{a}\right)T_1 = Q_{c_1} + \frac{Qabt}{4}$$

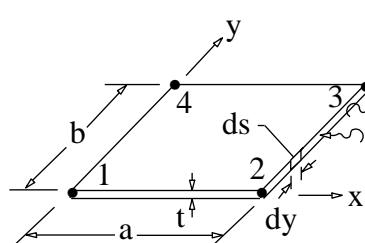
It can be shown that $Q_{c_1} = 0$ (no conduction across center of the plate), thus the temperature at the plate center is,

$$T_1 = \frac{3}{4} \frac{Qab}{k} \frac{1}{\left(\frac{a}{b} + \frac{b}{a}\right)}$$

Note that if $Q = k = a = b = 1$, the exact solution is 0.295 where T_1 is 0.375. But if use 4 elements, T_1 becomes 0.315, i.e., solution is improved with more elements.

PLATE WITH EDGE HEAT TRANSFER

If the plate has specified heating and convection along edge 2-3. Since the load vector due to edge heating is,



$$\{Q_q\} = \int_{S_2^{(e)}} q_s \{N\} ds$$

along edge $x = a$, $ds = tdy$ and

$$[N(x = a, y)] = \begin{bmatrix} 0 & 1 - \frac{y}{b} & \frac{y}{b} & 0 \end{bmatrix}$$

Then,

$$\{Q_q\} = \int_0^b q_s \begin{bmatrix} 0 \\ 1 - \frac{y}{b} \\ \frac{y}{b} \\ 0 \end{bmatrix} t dy = q_s b t \begin{bmatrix} 0 \\ \frac{1}{2} \\ \frac{1}{2} \\ 0 \end{bmatrix}$$

PLATE WITH EDGE HEAT TRANSFER

Similarly, the convection matrix corresponding to convection heat transfer along edge 2-3 can be derived,

$$\begin{aligned} [K_h] &= \int_{S_2^{(e)}} h \{N\} [N] ds \\ &= \int_0^b h \begin{Bmatrix} 0 \\ 1 - \frac{y}{b} \\ \frac{y}{b} \\ 0 \end{Bmatrix} \begin{Bmatrix} 0 & 1 - \frac{y}{b} & \frac{y}{b} & 0 \end{Bmatrix} t dy = h b t \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & \frac{1}{3} & \frac{1}{6} & 0 \\ 0 & \frac{1}{6} & \frac{1}{3} & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \end{aligned}$$

PLATE WITH EDGE HEAT TRANSFER

with the convection load vector,

$$\begin{aligned} \{Q_h\} &= \int_{S_2^{(e)}} h T_\infty \{N\} ds = \int_{S_2^{(e)}} h T_\infty \begin{Bmatrix} 0 \\ 1 - \frac{y}{b} \\ \frac{y}{b} \\ 0 \end{Bmatrix} t dy \\ &= h b t T_\infty \begin{Bmatrix} 0 \\ \frac{1}{2} \\ \frac{1}{2} \\ 0 \end{Bmatrix} \end{aligned}$$